epropelled°

Intelligent Power System iPS1500

Key Features

-P

One primary output (28 V) and two adjustable secondary outputs (12 V-14 V and 5 V-8 V)

(4)

High efficiencies of up to 93.5%, load dependent

Overcurrent and short circuit protection

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Onboard battery charging



Configurable EES parameters for different internal combustion engines



Real-time data monitoring for all voltages, currents, and temperatures

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LED indicators provide status for each output voltage and current



Software and hardware alerts to system controller

Fly Higher. Fly Longer. Fly Smarter.

Unmanned aerial vehicle (UAV) electronics continue to evolve as mission profiles become more demanding. System power designers are being challenged to provide more innovative power supply systems to improve efficiency, ensure reliability, reduce weight, minimize heat dissipation, and lower overall cost. New levels of energy and system-level efficiencies are also required to meet tomorrow's aviation needs.

Intelligent Power Systems

ePropelled intelligent power systems (iPS) are a complete power management solution for aviation applications. They convert the 3-phase sinusoidal AC voltage produced by a starter generator to tightly regulated DC voltage that can be used to power onboard avionics, servos, and payloads.

Designed to operate over a wide input range that varies with the speed of the starter generator or alternator, the iPS uses active rectification and switching regulation to supply the required steady DC output voltages.

These smart power systems also provide a wide array of real-time performance and operational data for a range of useful applications and analytics. The iPS monitors all input and output voltage, as well as current levels, and collects and reports that data via an integrated controller area network (CAN) interface. Custom applications can be created via our open application programming interface (API) and thresholds can be set for alerts and alarms based on specific applications and mission profiles.

Additionally, the iPS provides onshore DC power for all features including EES function, output power, and onboard battery charging.

ePropelled electronic engine starter (EES) feature is optional and can be used to drive the starter generator during the engine start sequence. Once the engine is up to speed, the iPS delivers the regulated voltages. If, for any reason, the starter generator stops working, an onboard battery (if connected) automatically engages to provide the required voltage for a limited time, dependent on the onboard battery size.

Battery Features

- > Onboard battery can provide power to outputs if 3-phase generator power is lost
- Onboard battery is charged when the unit is connected to 3-phase power or onshore DC power
- > Onboard battery can be used to power EES function
- > Onshore DC power for all features including EES function, output power, and onboard battery charging

Temperatures Monitored and Logged

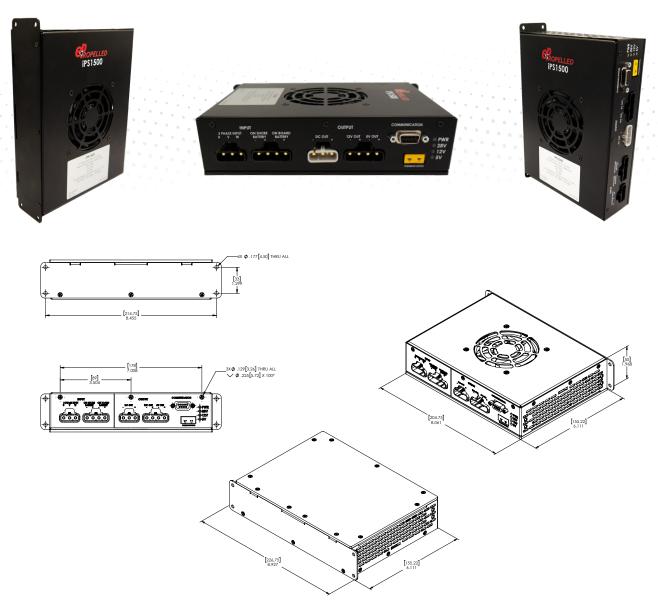
- Synchronous rectifier FETs
- DC converter FETs
- Output OR-ing FETs
- Starter generator via thermocouple

User-Configurable Parameters

- Conductor compensation voltage boost
- Alert and threshold settings
- RTDM settings
- CAN bus settings

Mounting Instructions

The figure below depicts the overall dimensions of the iPS chassis. Four holes are used for mounting the unit. Please note that weight and other details are provided in the technical specification table.



	iPS1500 SPECIFICAT			
Parameter	INPUT			
rarameter	Min	Max	Notes	
	25 V @ no load			
Input voltage range	50 V @ full load	95 V	RMS line-to-line	
Maximum total input power	1,662	2 W	At 40°C ambient	
Onboard battery voltage range	24 V	28 V	Battery type: 8S LiFePo4 or 7S LiPo	
Onshore DC voltage range	24 V	58 V	Input voltage below 33 V will not regulate the primary output	
Engine starter voltage range*	24 V	58 V	Supplied from onshore DC or onboard battery. (see note below)	
Engine starter current range	0 A	125 A	Peak/phase	
Start trigger voltage	2.3 V	5 V		
Start duration	0.25 s	10 s	5 seconds between attempts	
Parameter	OUTPUT			
	Min	Max	Notes	
Maximum total output power (continuous)	1,500) W	DC at 40°C ambient	
Primary output voltage	28	V	DC, max power = 1,500 W (53.6 A at 28 V	
Secondary output voltage 1	12 V	14 V	DC, max current = 20 A	
Secondary output voltage 2	5 V	8 V	DC, max current = 20 A	
Voltage regulation	±500	mV		
Voltage ripple P-P	500 r	mV		
Peak efficiency	93.5	1%	At 60% full load	
Onboard battery charging voltage	29.4	V	Battery type: 8S LiFePo4 or 7S LiPo	
Onboard battery charge current	3.33	A	Мах	
	Input undervoltage warning Output overvoltage warning			
Protection	*Output short circuit protection			
Trotection	Output overcurrent protection			
	Over temperature warning			
	Onboard battery switchover			
Parameter	MECHANICAL			
	Notes			
Dimensions	8.06" x	8.06" x 6.11" x 1.97" [204.8 mm x 155.2 mm x 50.0 mm]		
Weight		2.46 pounds (1,115 grams)		
Cooling		Fan cooled		
Ambient operating temperature		-26°F(-32°C) to 104°F (40°C) at 1,500 W		
Storage temperature		-40°F (-40°C) to 185°F (85°C)		
Ingress protection			IP20	

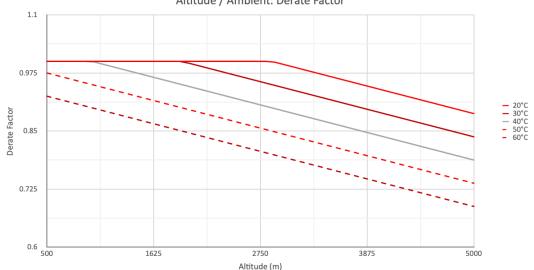
*Depending on the characteristics of the engine, the effectiveengine starter voltage range may be in a narrower range than specified. This value is only provided as an indication of the range possible and will be dependent on the specific internal combustion engine (ICE) the customer has specified.

*WARNING: When operating without an onboard battery, the unit has output short circuit protection. However, if a battery or a power supply is connected to the onboard battery terminals, the short circuit protection will force the unit into a switchover state when the output is shorted. This will cause damage to the circuit that is responsible for handling the switchover and it will void the warranty.

*WARNING The onboard battery must be fused with a 60 A fast blow in-line fuse. Failure to add the specified in-line fuse will result in damage to the unit and void the warranty.

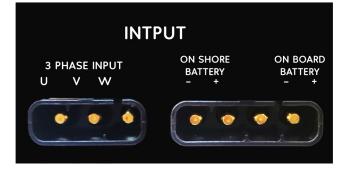
Derating with Increased Altitude

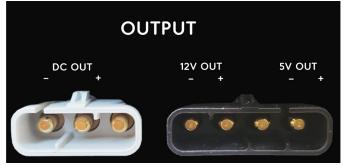
The derating factor for altitude is based on the loss of dielectric strength of the air as the density decrease with the altitude. The diagram below shows how the cooling efficiency changes with high altitude and ambient temperatures.



Altitude / Ambient: Derate Factor

	iPS	1500 PINOUT	
	-	nnector (Molex - 2018431030)	
		1-00011 (Molex - 2018410030)	
Pin	Label	Description	
1	U-LEG	3-phase connection U-LEG	
2	V-LEG	3-phase connection V-LEG	
3	W-LEG	3-phase connection W-LEG	
	DC Input Conn	ector (Molex - 2018431040)	
	Mating Cable - 68	1-00012 (Molex - 2018410040)	
Pin	Label	Description	
1	Onshore battery -	Ground. Return connection for the onshore battery.	
2	Onshore battery +	Positive connection for the onshore battery.	
3	Onboard battery -	Ground. Return connection for the onboard battery.	
4	Onboard battery +	Positive connection for the onboard battery.	
	Primary Output Co	onnector (Molex - 2018431031)	
	Mating Cable - 68	1-00013 (Molex - 2018410031)	
Pin	Label	Description	
1	DC out-	Ground. Return for 28 V primary output.	
2	N/A	Pin not used.	
3	DC out+	Positive 28 V primary output.	
		Connector (Molex -2018431041)	
		1-00014 (Molex - 2018410041)	
Pin	Label	Description	
1	12 V out -	Ground. Return for 12 V secondary output.	
2	12 V out +	Positive 12 V secondary output.	
3	5 V out -	Ground. Return for 5 V secondary output.	
4	5 V out +	Positive 5 V econdary output.	





Communications Connector (DB9-F)				
J7 Pin	DB9 Pin	Label	Description	
10	1	CAN-	CAN low.	
8	2	GND	Signal ground [for signals only].	
6	3	START, INPUT	Input "trigger" pin used to initiate a start attempt.	
4	4	GND	Signal ground [for signals only].	
2	5	TMS	Used only when reprograming device firmware.	
9	6	CAN+	CAN high.	
7	7	START, ENABLE	+3.3 V output used for triggering the starter.	
5	8	BB SIGNAL, OUTPUT	Output signal indicating that the unit is operating on onboard battery power.	
3	9	ТСК	Used only when reprograming device firmware.	
1	N/A	Not used		
			Thermocouple Input	
Pin		Label	Pin Description	
+ TC+		TC+	Positive K-type thermocouple input.	
- TC-		TC-	Negative K-type thermocouple input.	

Note: All grounds are internally connected (this includes both power and signal ground).



Recommended Applications

- Aircraft power systems
- Unmanned vehicle power systems
- Power conditioning
- Stand-alone power systems (SAPS) for remote area power supply
- Voltage regulation in the renewable power generation system

Assembled in USA

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